



4-6 K Ball Aerospace Long Life Space Cryocoolers

Dave Glaister and Willy Gully

**(303) 939-5842
dglaister@ball.com**

**TPF Science, Technology and Design Expo
October 14-16, 2003**



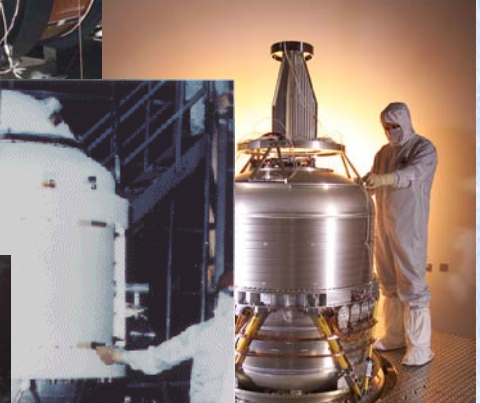
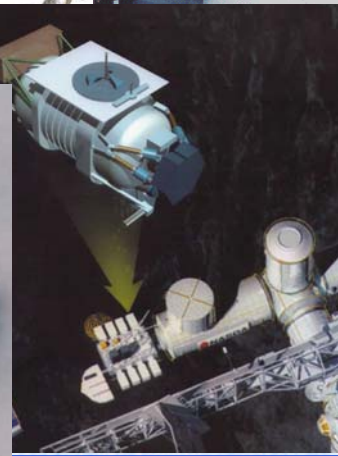
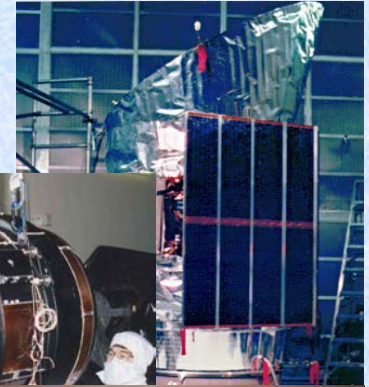
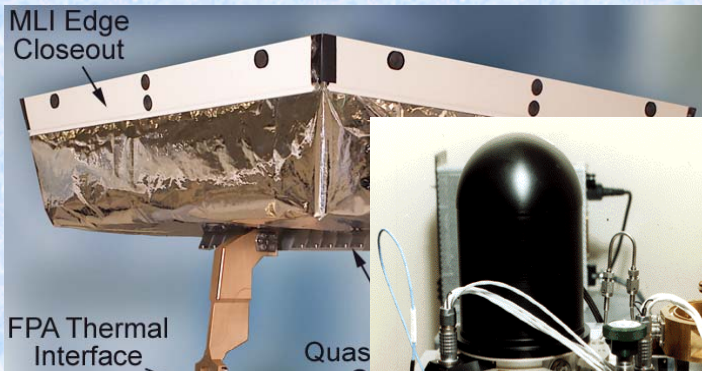
Ball Cryocooler Strengths

- **Cryogenic Systems Expertise** – For over 40 years, Ball has done and continues to do more space cryogenic work than another organization in the world
- **Low Temperature Specialization** – Over a decade of <20 K space cryocoolers culminating in the 4-10 K cryocooler
- **Multi-Stage Expertise** – Over a decade of multi-stage flight cooler that are optimum for dual loads and low temperatures
- **Build-to-Print Systems** – Build-to-Print SB160 and SB235E Coolers and E200 and E300 Electronics
- **Strong Emphasis on System Integration** – Unique integration features of the Ball cryocoolers minimize integration risk, thus, significantly reducing the overall program risk



Ball Cryocooler Expertise Based on Specialization in Low-Temp Cryogenics

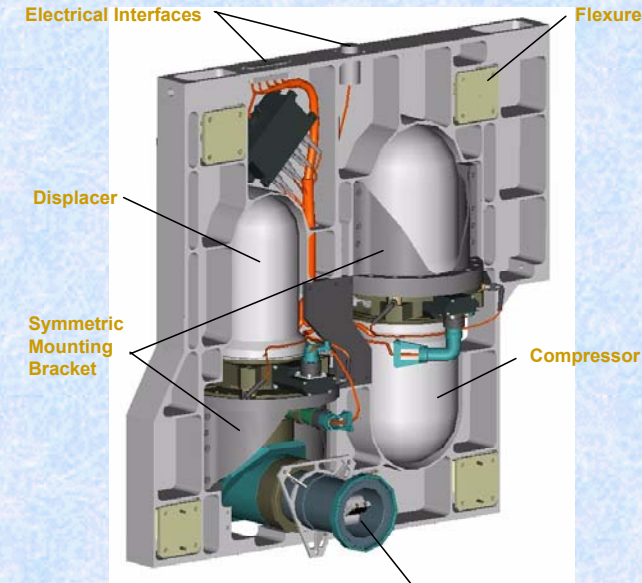
- Ball Aerospace has experience with more than:
 - 40 years on cryogenic spaceflight instruments
 - 18 years on cryocoolers
 - 11 years on multi-stage Stirling coolers
 - 150 cryogenic space flights





Ball Cryogenic Systems Engineering

- Historically, Cryogenic Systems are Relatively Difficult to Develop
 - Space cryogenics is very personnel dependent
 - Industry expertise base is declining
- Ball Specializes in Cryogenic Systems
 - Built space cryogenic systems for nearly 50 years
 - Over 150 cryogenic system flights
 - Over 2000 man-years of cryogenic expertise
 - More than 80 people currently supporting cryogenic programs
- Ball Builds and Integrates Cooling Sources
 - Mechanical cryocoolers
 - Dewars (cryostats)
 - Cryoradiators
- Ball Designs and Builds Cryogenic Instruments





Ball Cryocoolers Cover Wide Range of Applications with Multiple Technologies

- Multiple Long Life Cooler Product Lines
 - Each optimum for different application envelopes
 - Each based on proven long life designs
- Stirling Coolers
 - Very compact and power efficient
 - In build for over a decade
- Joule-Thomson Coolers
 - Inherent load leveling capability
 - Optimum for stable temperature over distributed cooling interface
 - In build for over a decade
- Hybrid Coolers
 - Combines advantages of Stirlings and J-Ts
 - Optimum for low temp (<10 K) applications
- Optical Cooler
 - Very long life (no moving parts)
 - Very compact, light, with tight integration
 - Zero vibration, zero EMI
 - Low-cost manufacture





Flight Qualified 1, 2, and 3-Stage Ball Stirling Coolers with Significant Life Testing

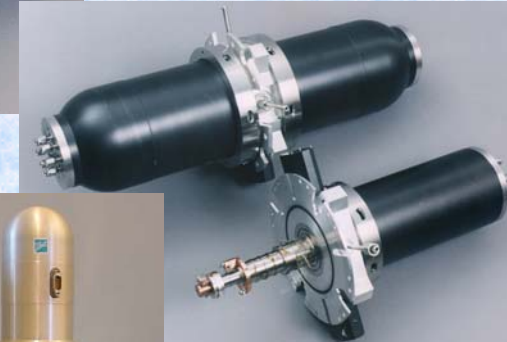
- Long Life, Multi-Stage Stirlings since 1989
- Flight Qualified 1, 2, and 3-Stage Coolers
- Flight Qualified Drive Electronics
- 1-Stage Cooler (SB160 or HIRDLS)
 - Optimum at 60 K and above
 - 2 Flight Units delivered for NASA Aura
 - ~5,000 hrs. on HIRDLS Coolers
 - ~27,000 hrs. on SA160
- Next Gen 2-Stage Cooler (SB235)
 - Build-to-print, qualified cooler
 - Optimum for FPA & Optics cooling
 - Mass and power efficient, high capacity, producible cooler
- 3-Stage Cooler (SB335)
 - Optimum for <30 K
 - ~17,000 hrs. on qualified SB335
 - SB315 precooler for Ball ACTDP 6 K Cooler
- Heritage 2-Stage Cooler (SB230)
 - ~20,000 hrs. on SB230
 - (11,000 hrs. on displacer)



1-Stage HIRDLS



2-Stage 30 K



3-Stage 35/60 K



2-Stage
SB235



SB235 Cryocooler

○ SB235 Stirling Cryocooler

- Ball's 3rd Generation Multi-stage 35 K Cooler
- Designed for producibility with increased efficiency & capacity
- Met SBIRS Low Coude' (SDR) Track rqmts.
- Fight qualified unit entering life test

○ Performance improvements over SOA

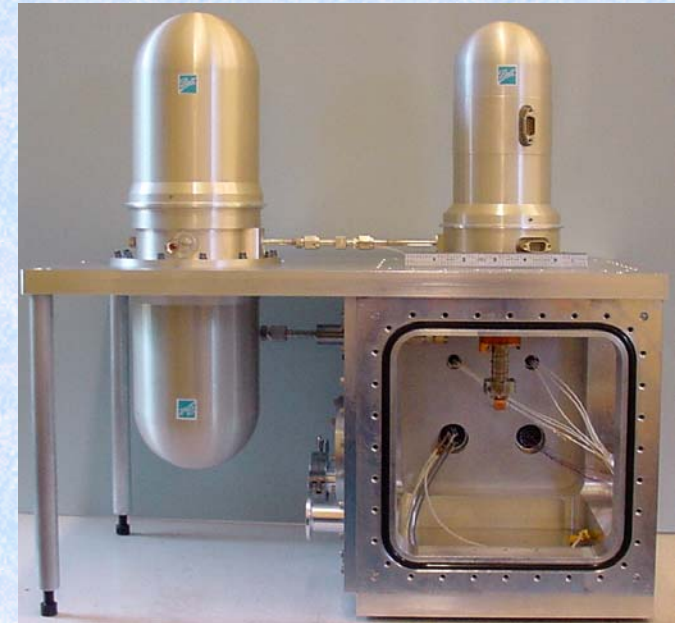
- 2-3 X cooler capacity
- ~40% increased power efficiency
- ~50% increased mass and volume efficiency

○ Producibility/Cost improvements

- 30% reduced parts count
- 75% reduced match-machining
- Reduced complexity & simplified assembly

○ Performance

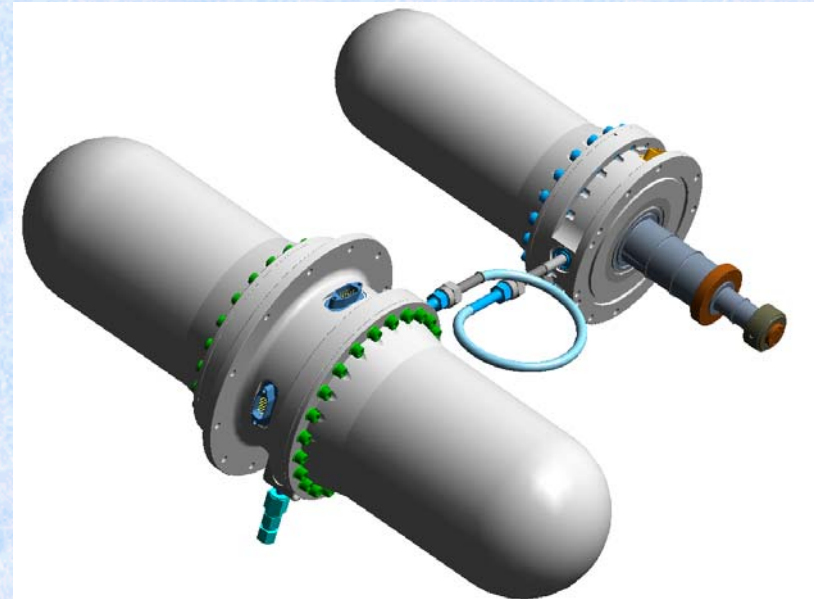
- 0.5 W @ 40 K & 3.5 W @ 110 K for 90 W motor
- 99% reliability at 10 years
- 10.5 kg mass





Build-To-Print Ball SB235E Cryocooler System

- Higher capacity model derivative of qualified SB235
 - Qualified SB235 unit entering life test
 - Baselined IPDR cryocooler for the Raytheon SBIRS Low Track Sensor
 - Baselined for NASA/JPL Advanced Cryocooler Technology Development Program (ACTDP)
- Build-To-Print Cooler
- Performance
 - Higher mid-stage capacity than SB235
 - 1.2 W @ 40 K & 12.0 W @ 110 K for 171 W motor
 - 3.2 W @ 56K & 14.6 W @ 120K for 205 W motor
 - 2.6 W @ 56K & 11.7 W @ 120K for 176 W motor
 - 99% reliability at 10 years
 - 12.9 kg mass





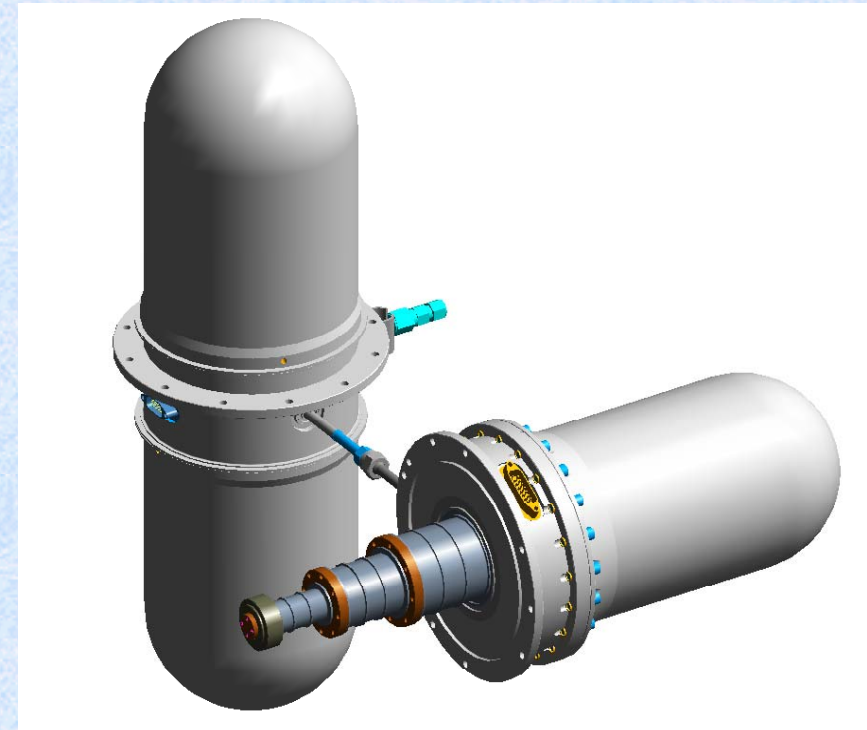
SB315 Cryocooler

○ 3-Stage Cooler for cooling below 15 K

- Highly leveraged of SB235 and SB335 coolers
 - High capacity from SB235
 - 3-Stage cold tip from SB335
- ACTDP Precooler for 4-6 K J-T cooler
- Engineering Model in build on NASA/JPL's ACTDP program.

○ Performance

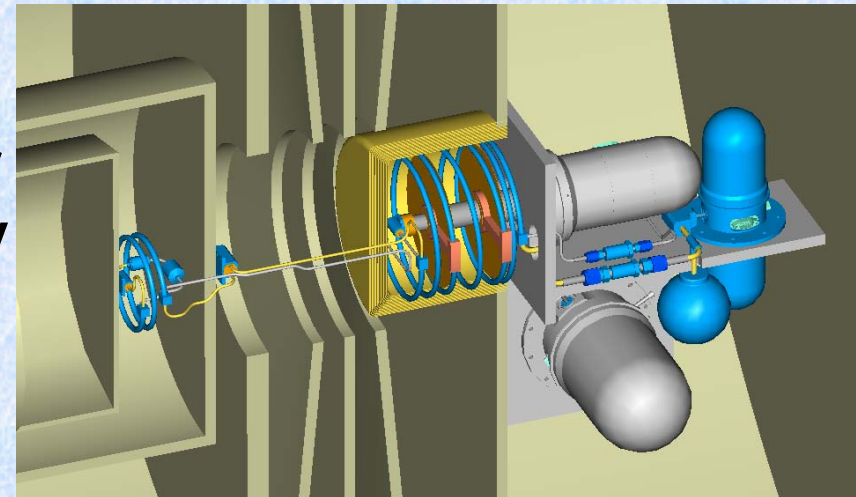
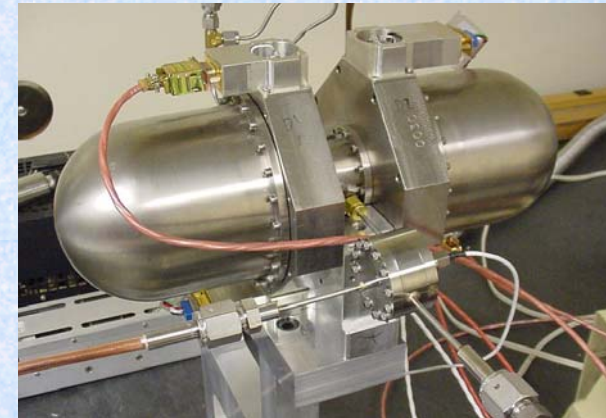
- Higher mid-stage capacity than SB235
- 0.3 W @ 15 K & 1.0 W @ 40 K & 2.0 W at 180 K for 180 W motor
- 99% reliability at 10 years





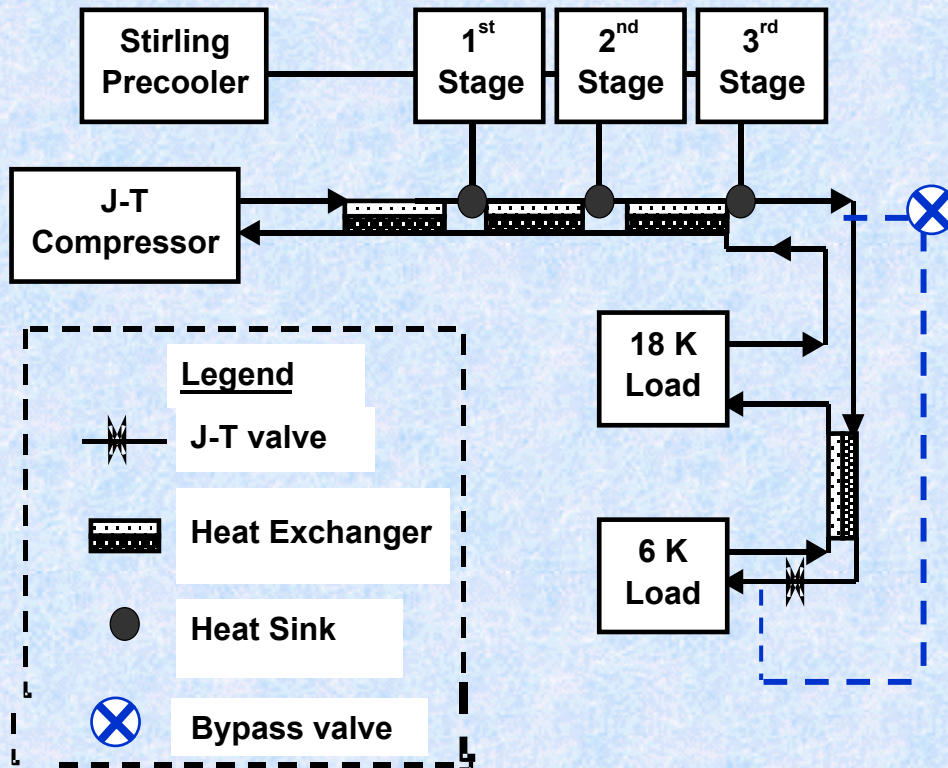
Ball 4-10 K (ACTDP) Cryocooler

- ACTDP (Advanced Cryocooler Technology Development Program)
 - NASA/JPL 6 K cooler development for JWST, TPF, and Con-X programs
 - 3 years culminating in Engineering Model
 - DoD application to 10 K VLWIR systems
- Cooler design is at Post-PDR level
 - Design detailed down to piece-part level
 - Leveraged off previous Ball Stirling and J-T hardware
- Performance
 - 30 mW @ 6 K (or 20 mW @ 4 K) & 150 mW @ 18 K for 125 W motor
 - On-orbit 2X load change capability
 - 5X change by swapping J-T valve
 - 30 kg
 - >20 m remote heat transport





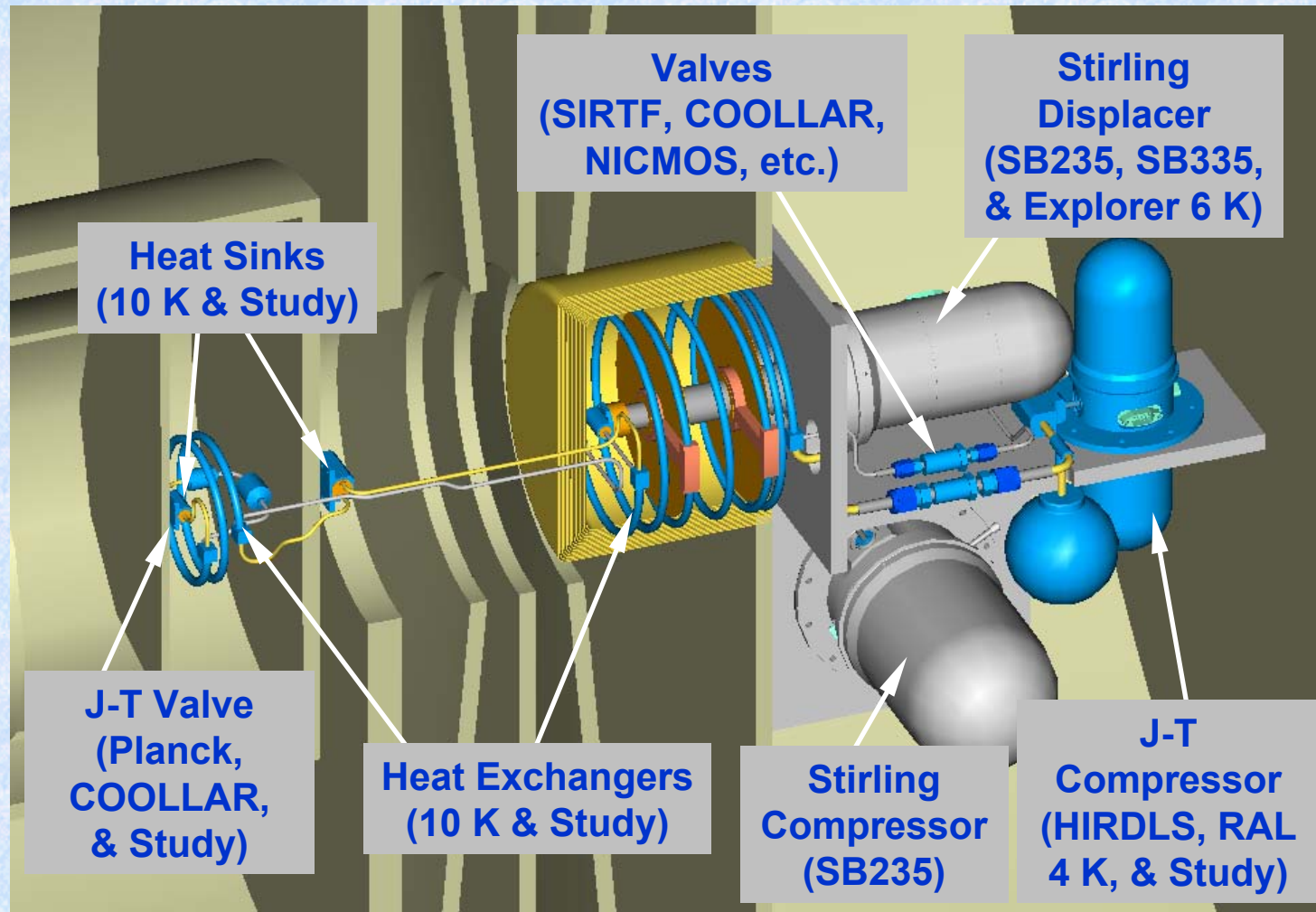
Operational Fundamentals



- Optimum combination of two cooling cycles
 - Proven Stirling efficiency to cool from ambient to 15 K
 - J-T recuperative efficiency to cool below 15 K
- Two methods of cooling at the two loads
 - J-T directly cools 6 K
 - J-T provides transport/ circulation of Stirling 18 K cooling after 6 K
- Bypass added to expedite cooldown

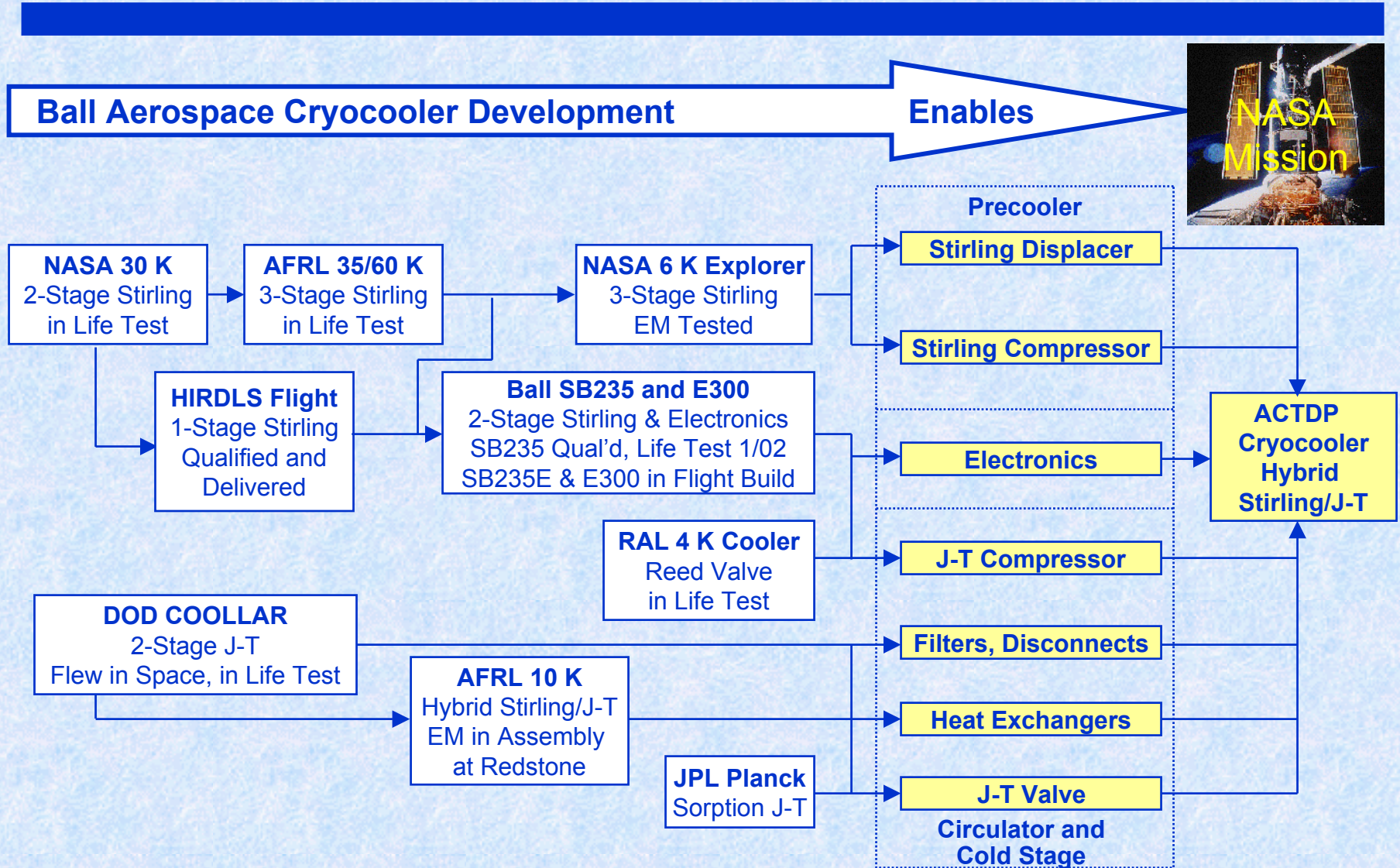


All Cooler Components Proven in Test on Study Phase and Previous Programs





ACTDP Cryocooler Highly Leveraged Off Previous Development

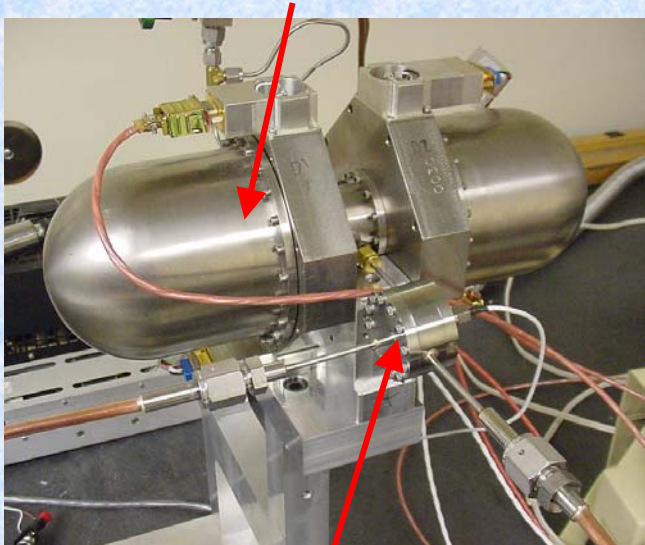




All Technology Components Verified in Test

- Culminated in successful 4 and 6 K system tests (35 mW @ 6 K, 21 @ 5 K, 12 @ 4 K, 0 @ 3.4 K)

Long Life, Linear Compressor

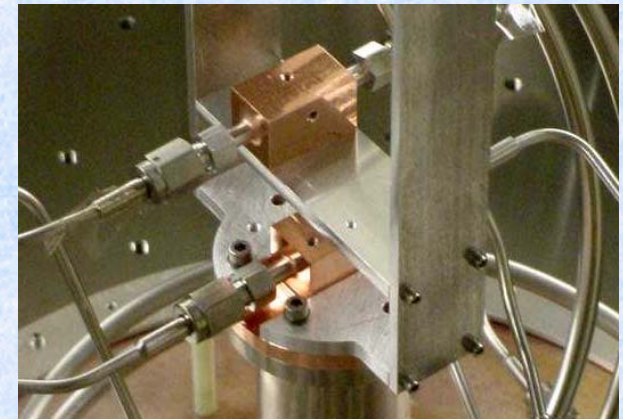


Reed Valve

Heat Exchangers



Heat Sinks



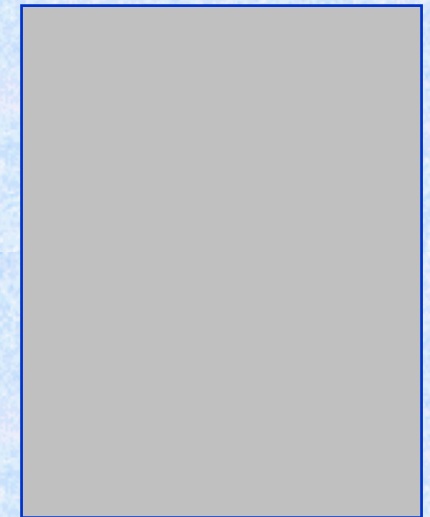
J-T Valves





ACTDP Cooler Design Optimized for Critical System Integration

- Remote heat transport (no moving cold head parts)
 - No moving cold head parts can be deployed and located >20 m from compressor suite
- No cryogenic radiator required or needed
 - Equally applicable to Earth and non-Earth orbit missions
 - No stray light or thermal back load on instrument
- Mature thermal/mechanical integration system
 - Including structural supports/thermal insulation
 - Conceptual integration into JWST, Con-X, and other program designs
- No 0-g concerns or 1-g testing limitations
 - Single phase, gaseous working fluid
- Independent control of 6 and 18 K heat loads
 - Real-time, on-orbit capability to adjust $\pm 2x$ to actual conditions



Ball Aerospace Cooler
Integrated into Con-X XMS